



Feedback Loop on a large scale quadrupole prototype

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Overview

- Brief summary of the last presentation in London
- Transfer of the former studies to a large scale prototype :
 - *Description of the prototype with appropriate actuators.*
 - *Results of the active vibration reduction*
- Robustness : Development of one frequency tracking in real time
- Technology and location of the instrumentation
- Conclusions

Collaboration with 



The Spectrum of disturbances is not a white noise
(ground motion, acoustic noise...)

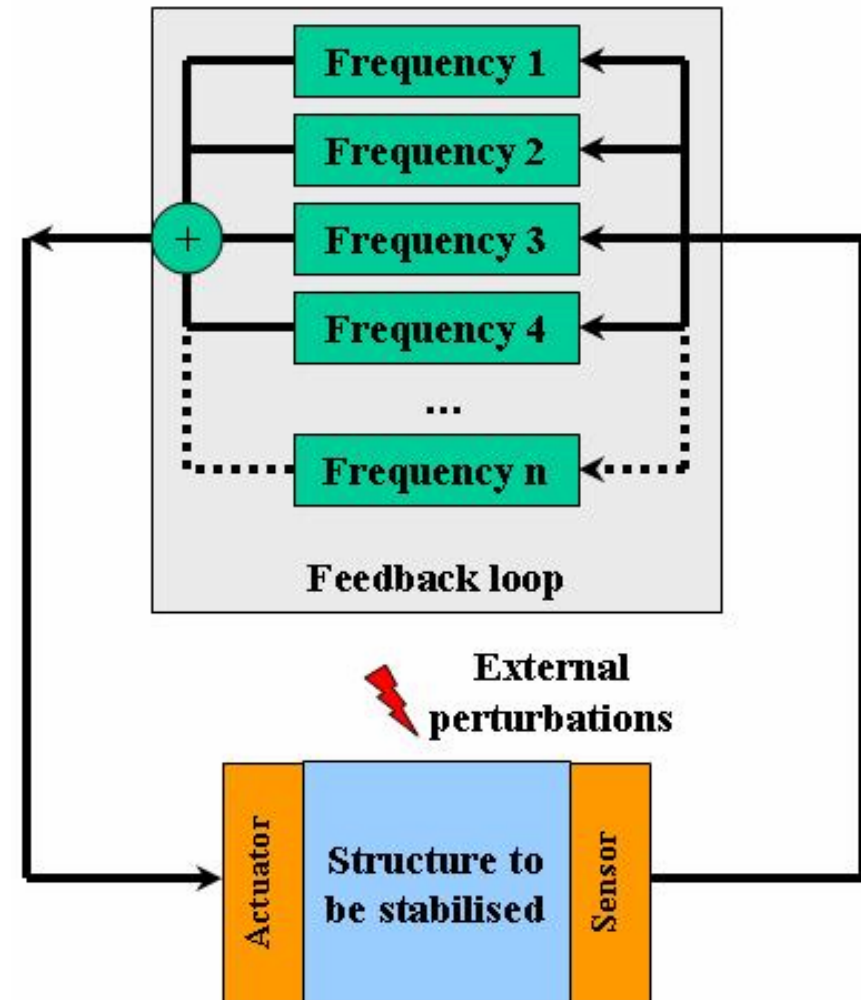
↓ Filtering by the mechanical structure

The global effect of the disturbance contains :

- Some frequencies which are amplified
- All frequencies are independent

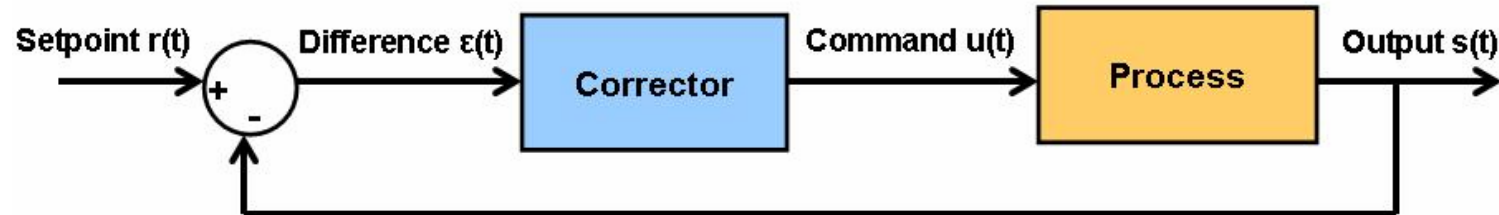
↓

Strategy : to control independently every main frequency

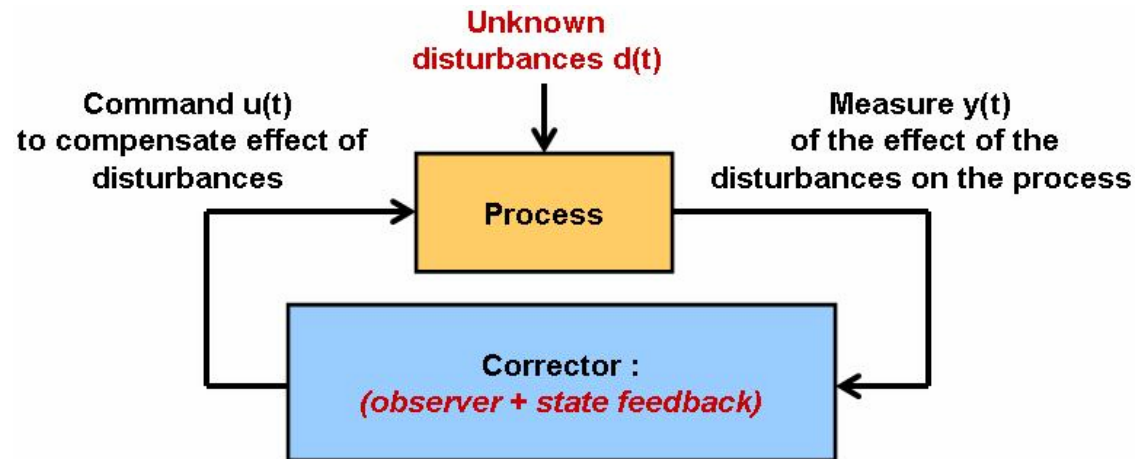


Originalities of the method : the algorithm

➔ Usually, a classic algorithm (ex : PID), depends on the model of the process :



➔ **It is not a classic algorithm**, it is a compensation of the disturbance :
(without knowing the model of the process, only its behaviour at certain frequencies)





Originalities of the method : the signal processing

→ A non linear problem :

$$f(t) = \alpha \sin(\omega t + \varphi) \quad (\text{the force to be computed})$$

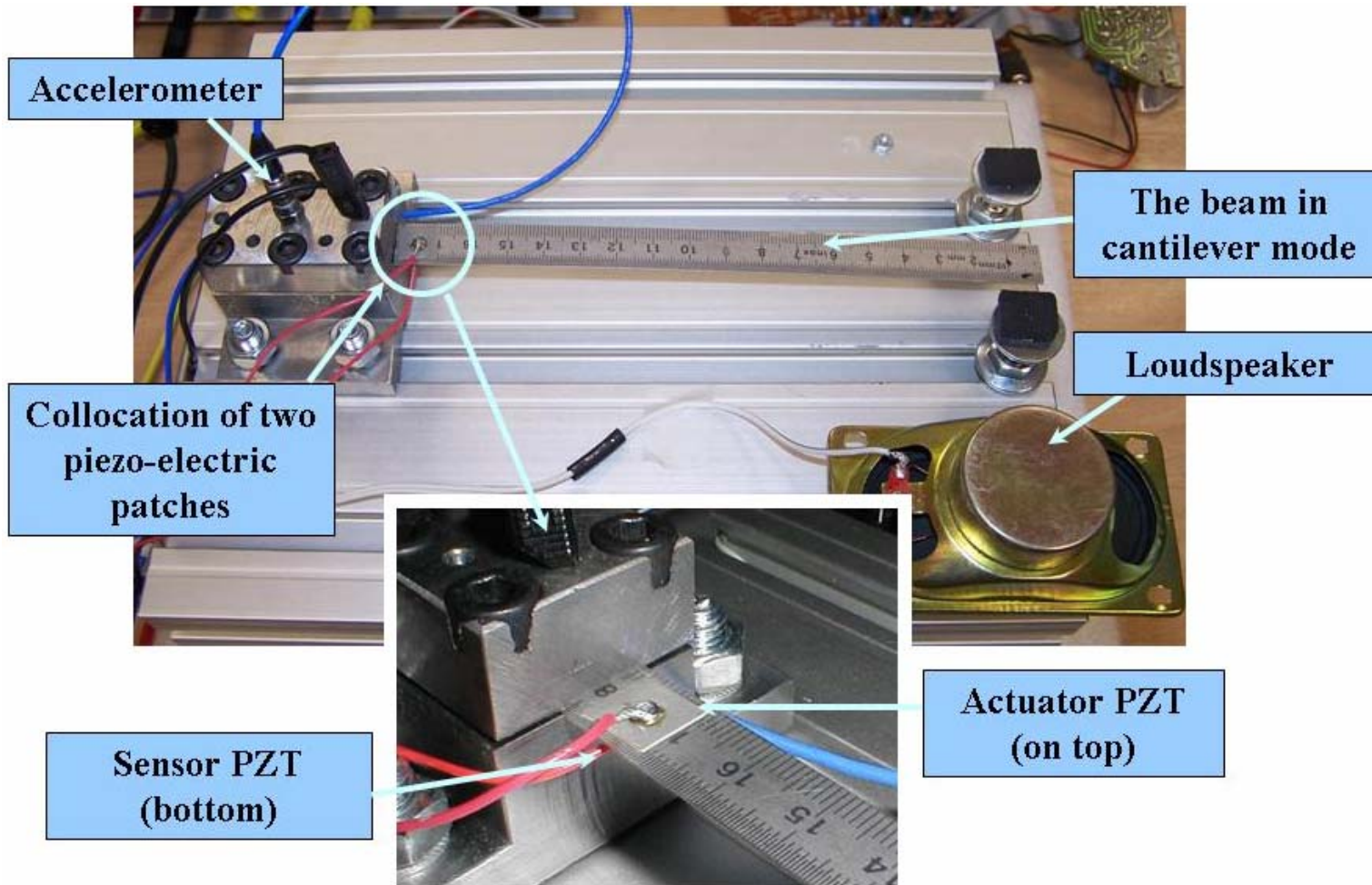
→ Decompose each resonance as a weighted combination of sine and cosine :
(*measurement, disturbance, excitation*)

$$f(t) = \alpha \sin(\omega t + \varphi) \Rightarrow f(t) = f_s \sin(\omega t) + f_c \cos(\omega t)$$

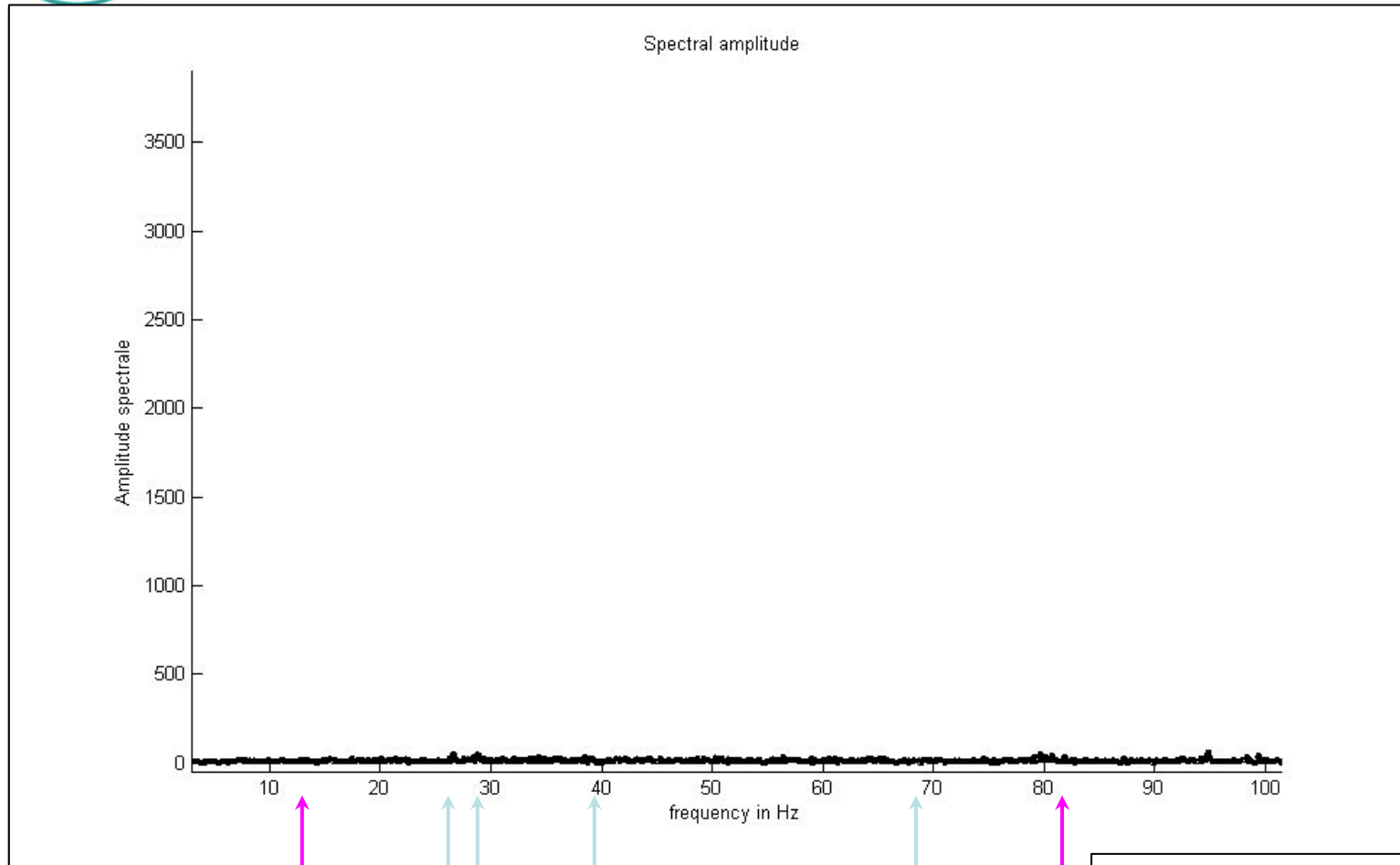
where:

$$\begin{aligned} f_s &= \alpha \cos(\varphi) \\ f_c &= \alpha \sin(\varphi) \end{aligned}$$

Description of the prototype :



Rejection of 6 resonances : (without and with rejection)



Resonances of : *-beam*
-support



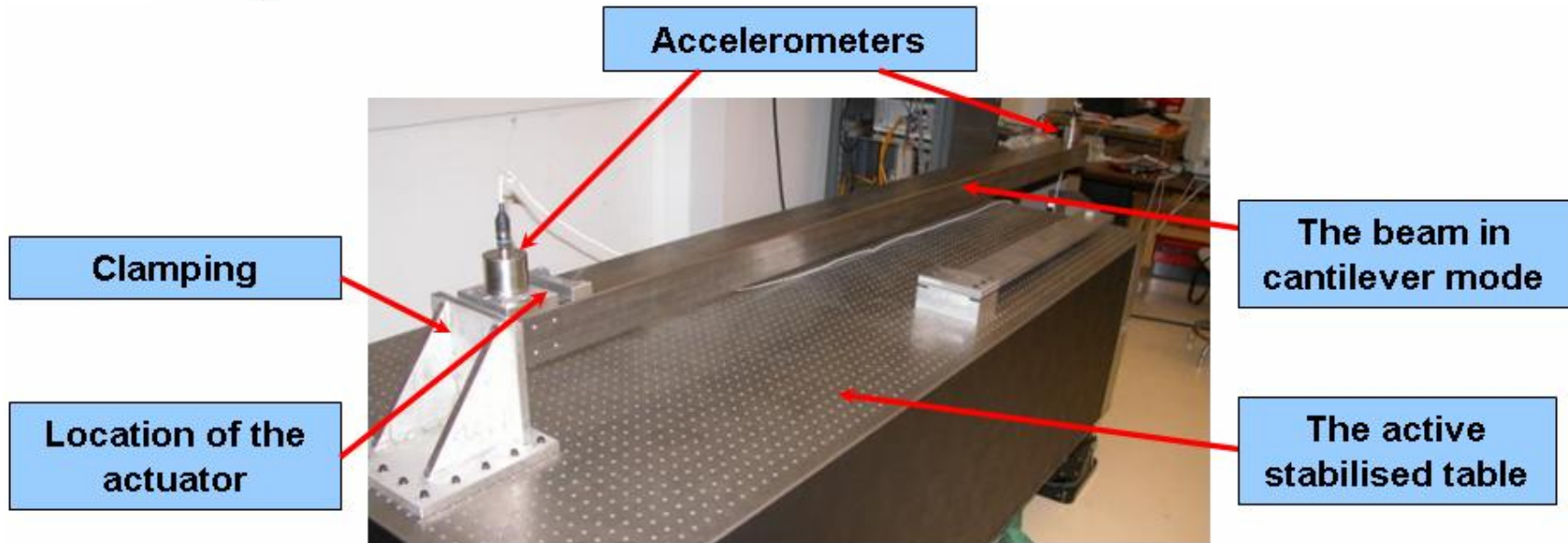
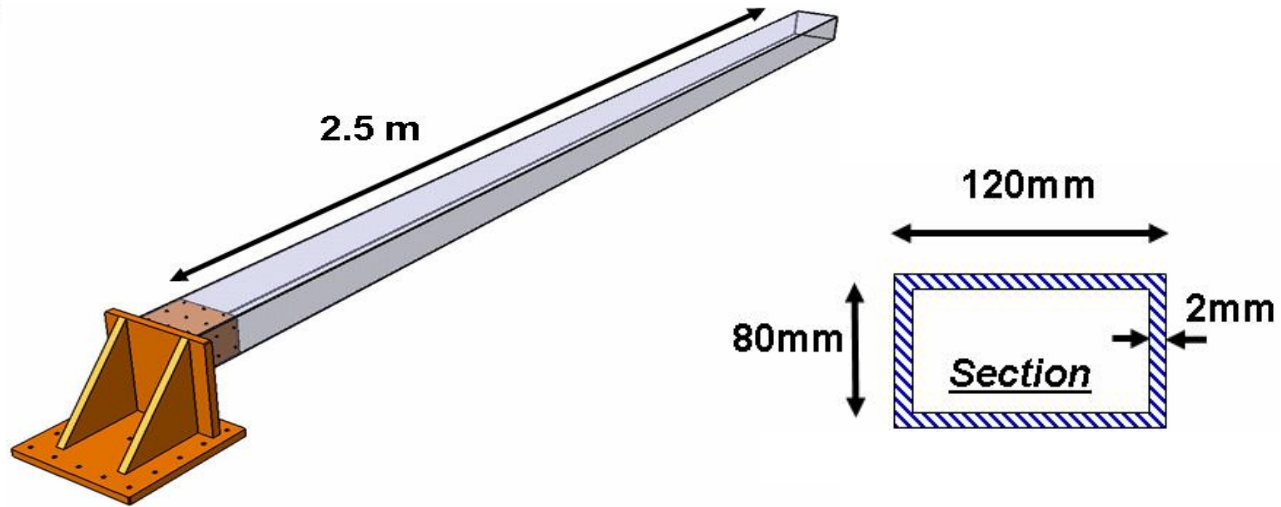
Why a new prototype?

- To validate the algorithm on a large scale prototype, whose size, boundary conditions and eigen frequencies are similar to the final focus.
- To validate the micro-computing which :
 - manages noisy low signal with very high resolution
 - computes the appropriate control of the feed-back in a limited time.
- To validate sensors and actuators which are performant, compatible and adapted to the final focus.
- To validate the developed simulation for the prediction.

➔ Movement of a linear mechanical structure < ground motion

The mock-up :

A large scale prototype





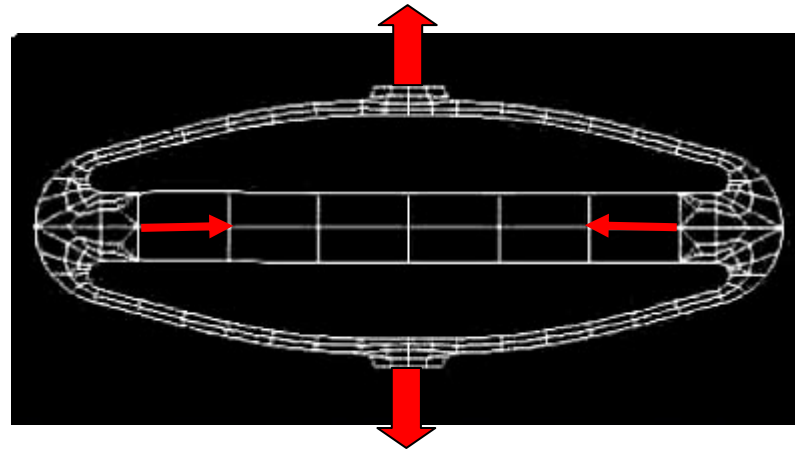
The actuator : description

A large scale prototype

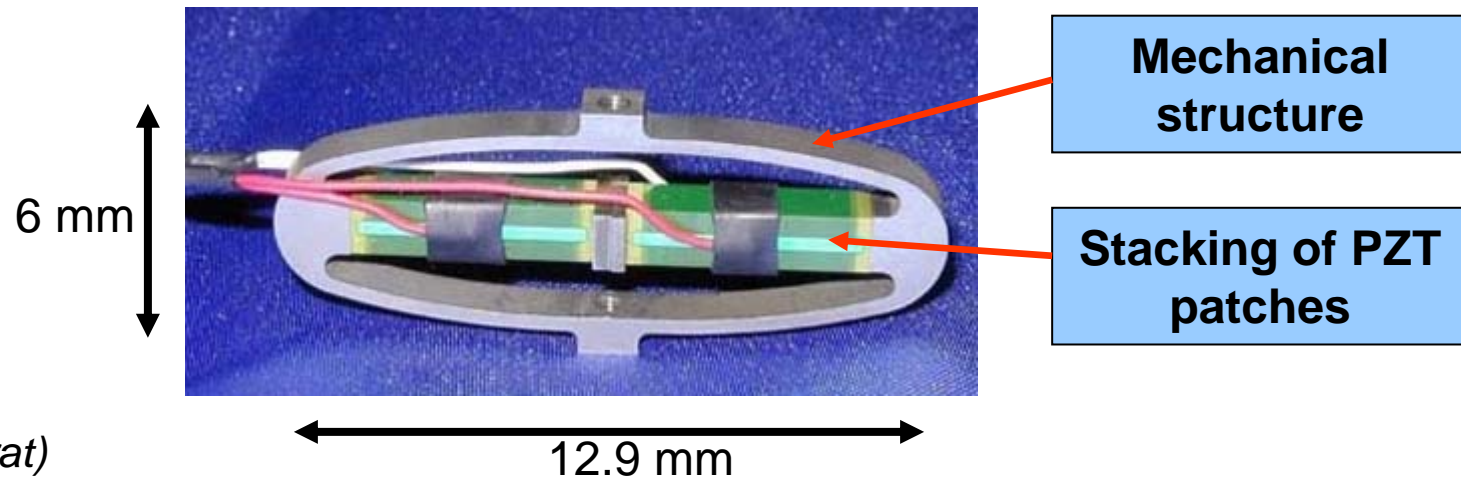
Force = 19.3 N

**Maximal displacement
= 27,8 μm**

Resolution = 0,28 nm



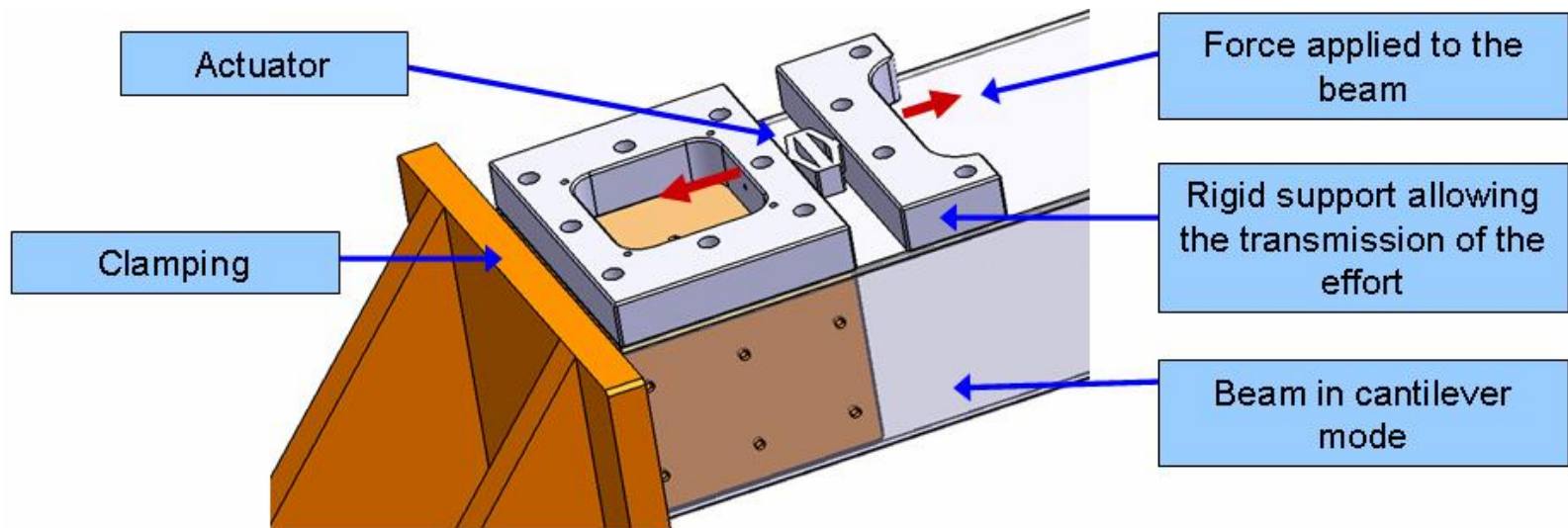
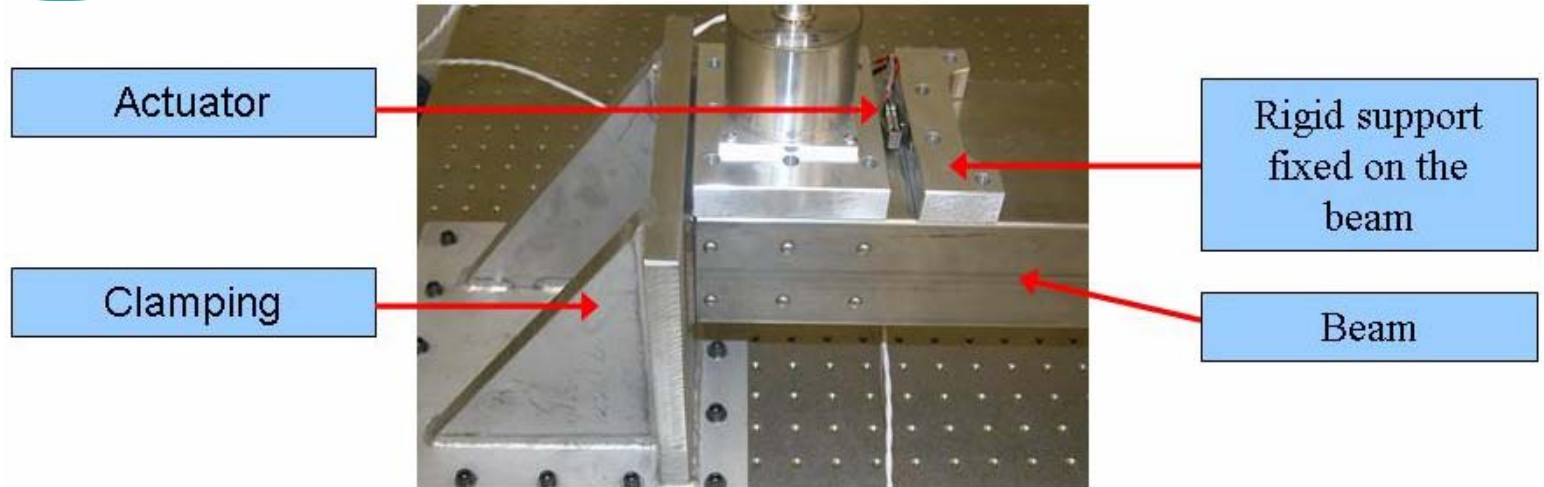
The deformation of the PZT patches is amplified by the mechanical structure



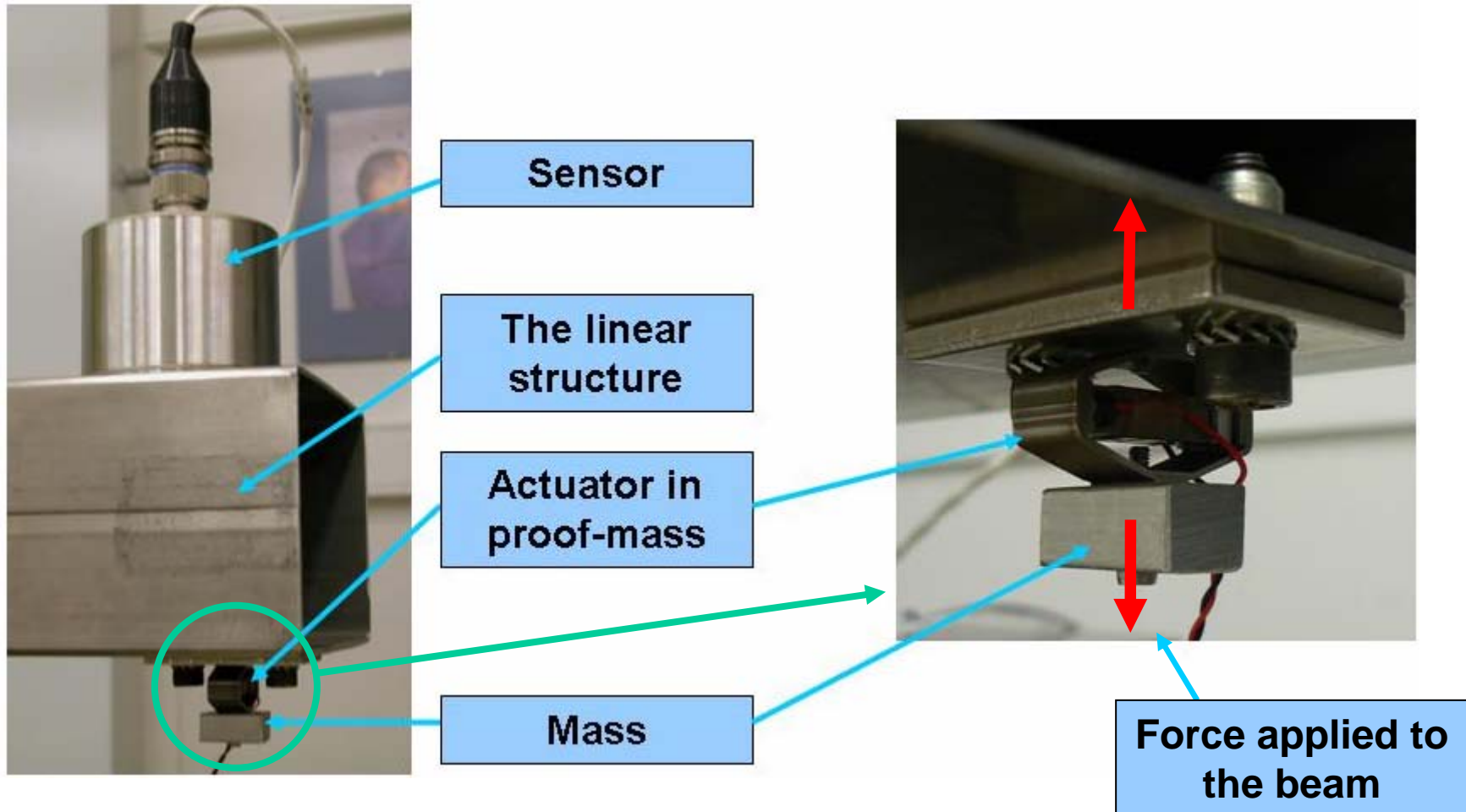
(Produced by Cedrat)



The actuator applies a force in flexion

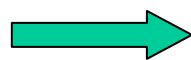
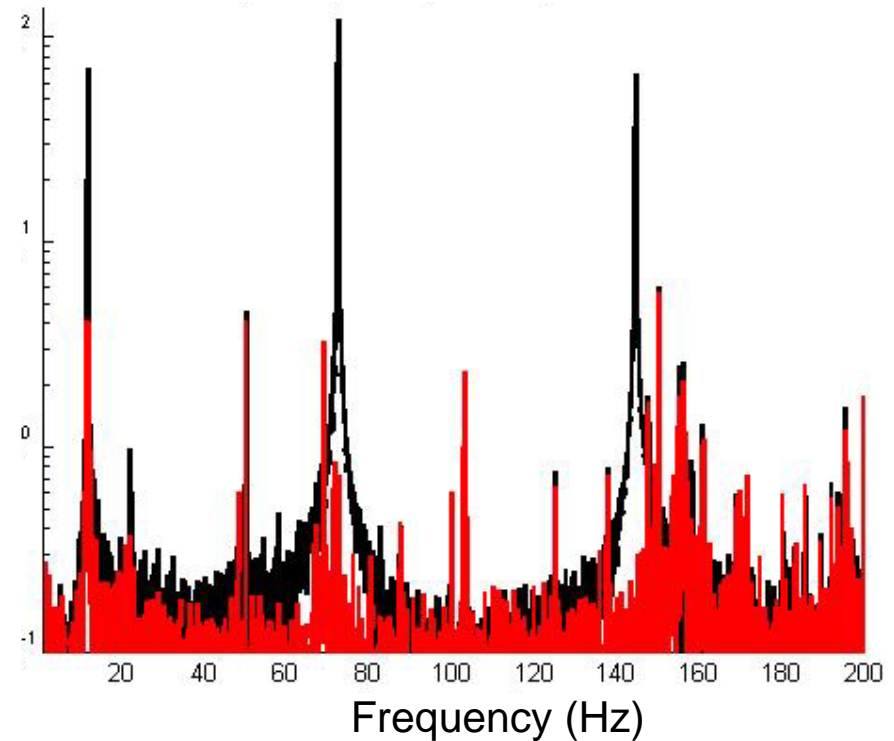
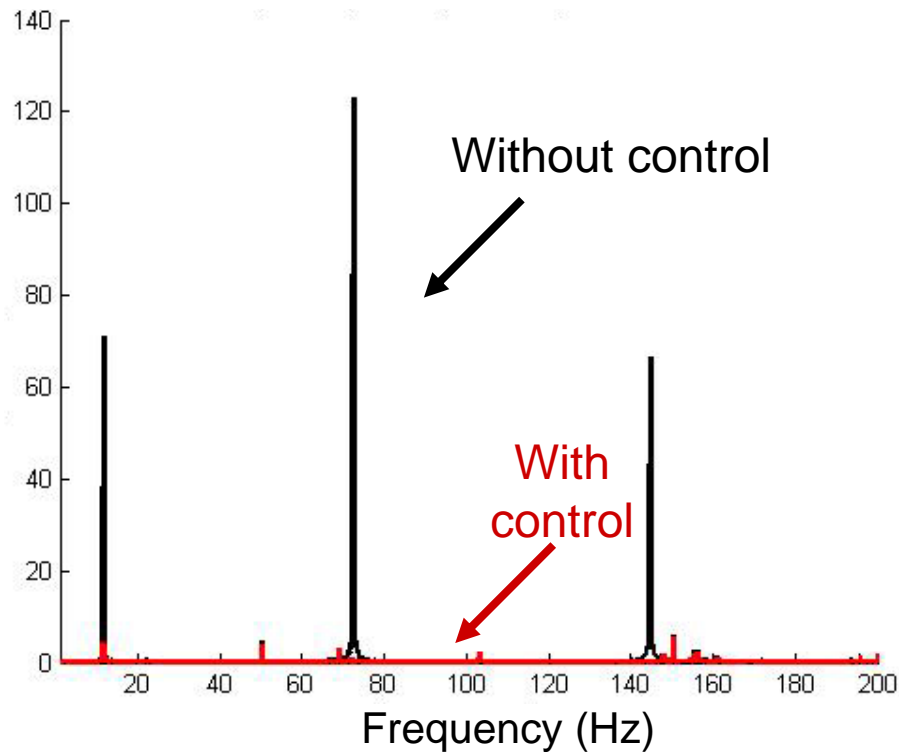


A large scale prototype
The actuator applies a force in “proof-mass”



Results : rejection of 3 fixed frequencies of disturbances

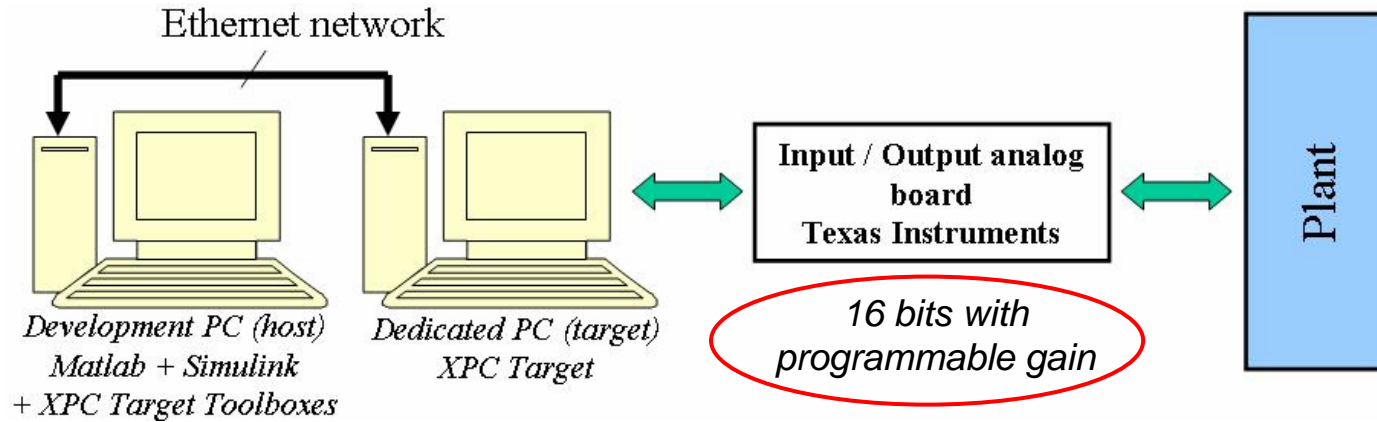
Spectral amplitude of measured signals



Efficient vibration rejection

Limitations :

Current configuration : Fast prototyping with XPC Target in a dedicated PC



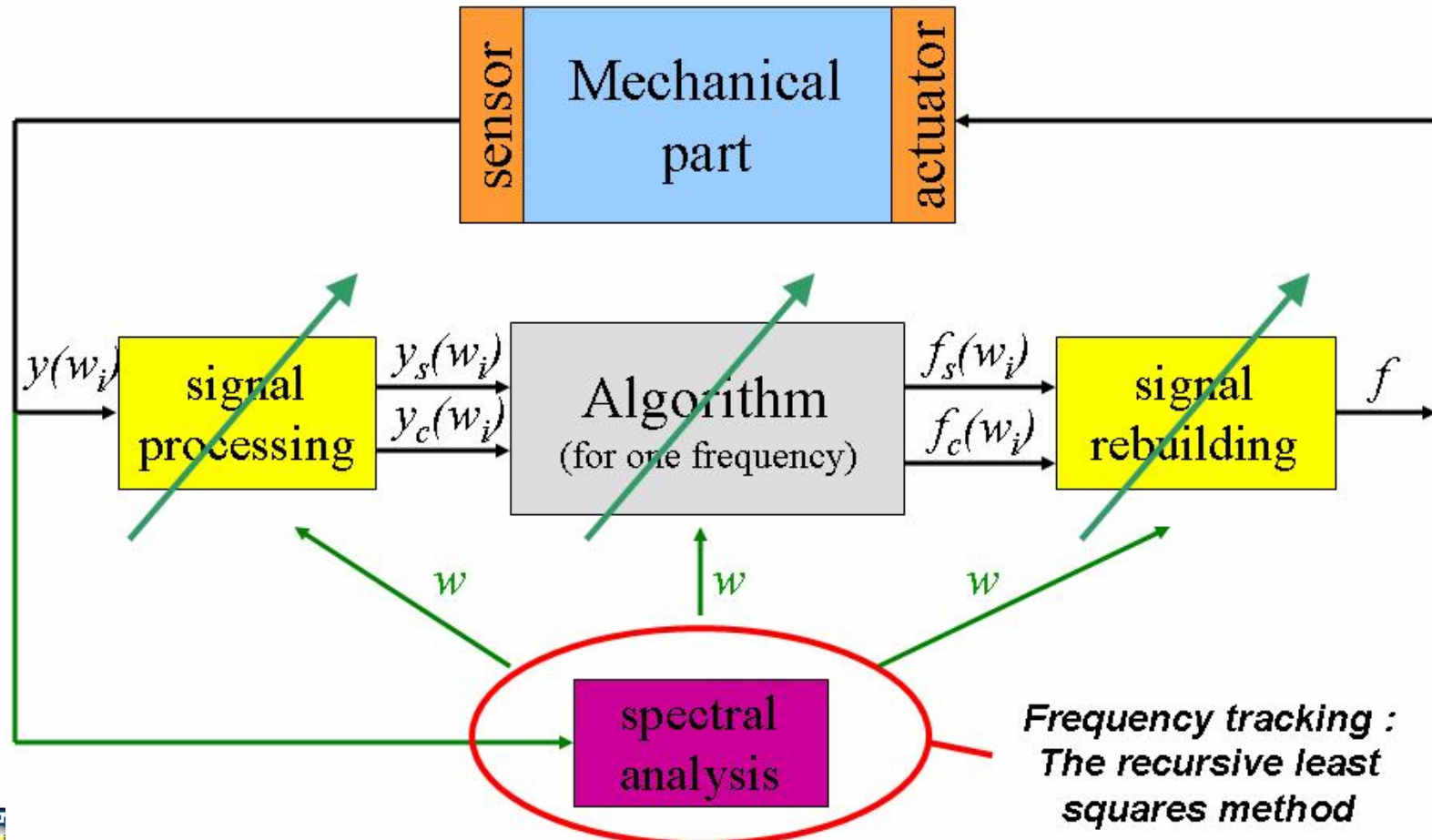
Configuration in test : DSP (“Digital signal processing”) of ProDAQ



- 24 bits resolution
- Programmable gain...
- Adapted electronic for vibrations

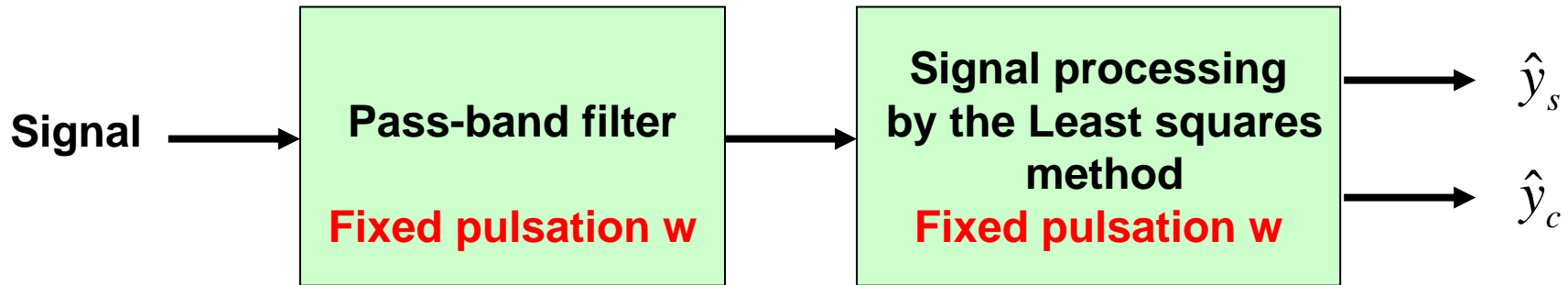
➔ **possibility to reduce to nanometer scale**

- As each frequency is rejected independently, the robustness depends on the estimation of the real value of the disturbance frequency :



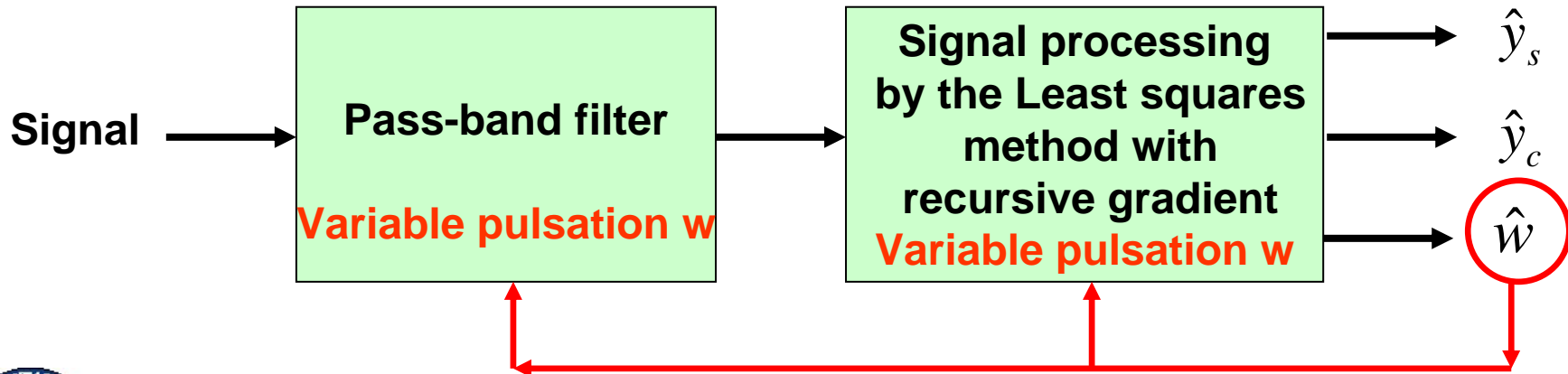
Initial Method of “signal processing” :

$$\hat{y}_i = \hat{y}_s \cdot \sin(\omega t_i) + \hat{y}_c \cdot \cos(\omega t_i)$$



Objective:

$$\hat{y}_i = \hat{y}_s \cdot \sin(\hat{\omega} t_i) + \hat{y}_c \cdot \cos(\hat{\omega} t_i)$$



Recursive least squares method

Considering the measurement has the following form:

$$y(t_i) = y_s \cdot \sin(\omega t_i) + y_c \cdot \cos(\omega t_i)$$

The criterion to be minimized :

$$J(y_s, y_c) = \frac{1}{N} \cdot \sum_1^N (y_i - y(t_i))^2 \quad (N : \text{number of samples})$$

Matrix form:

$$\begin{matrix}
 & \begin{matrix} \left(\begin{matrix} y_1 \\ y_2 \\ \dots \\ y_N \end{matrix} \right) \\
 \swarrow Y & = & \begin{matrix} \left(\begin{matrix} \sin \omega t_1 & \cos \omega t_1 \\ \sin \omega t_2 & \cos \omega t_2 \\ \dots & \dots \\ \sin \omega t_N & \cos \omega t_N \end{matrix} \right) \\
 & \searrow H & \begin{matrix} \left(\begin{matrix} y_s \\ y_c \end{matrix} \right) \\
 & & \searrow \hat{M}
 \end{matrix}
 \end{matrix}$$

The matrix M which minimizes the criterion :

$$\hat{M} = (H^T \cdot H)^{-1} \cdot H^T \cdot Y$$



Recursive least squares method

Frequency tracking in real time

Minimizing the criterion J corresponds to minimizing its derivative by the variables to be estimated :

$$\frac{\partial J}{\partial \hat{y}_s} = 0$$

$$\frac{\partial J}{\partial \hat{y}_c} = 0$$

$$\frac{\partial J}{\partial \hat{w}} = 0$$

Recursivity only for w

Which gives :

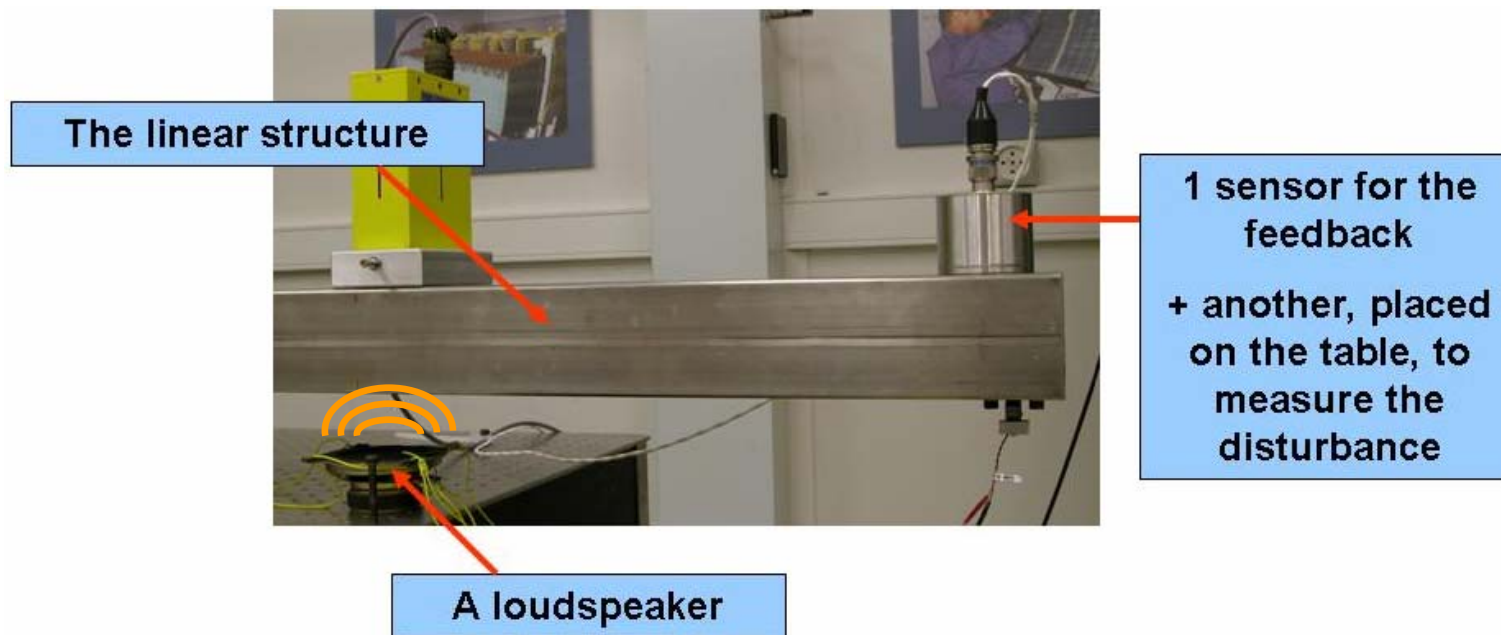
$$\hat{w}_{j+1} = \hat{w}_j - \lambda \frac{\partial J}{\partial \hat{w}}$$

(where λ is the dynamics of the recursivity)

The ending criterion of the recursivity :

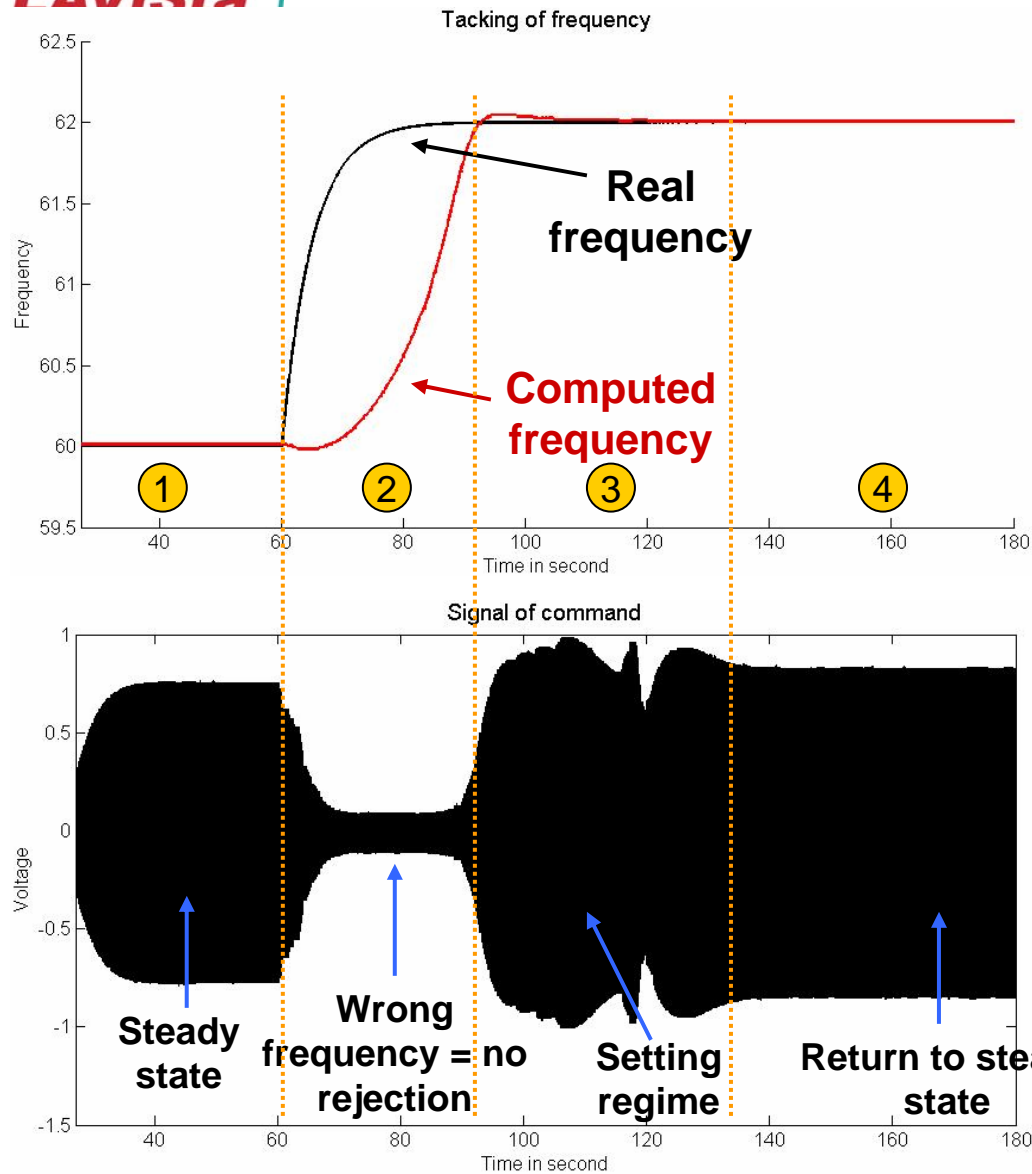
$$\frac{J(\hat{w}_{(j+1)}) - J(\hat{w}_{(j)})}{J(\hat{w}_{(j)})} < \varepsilon$$

- External disturbance simulation with a step frequency function (response of a 1st order process) :

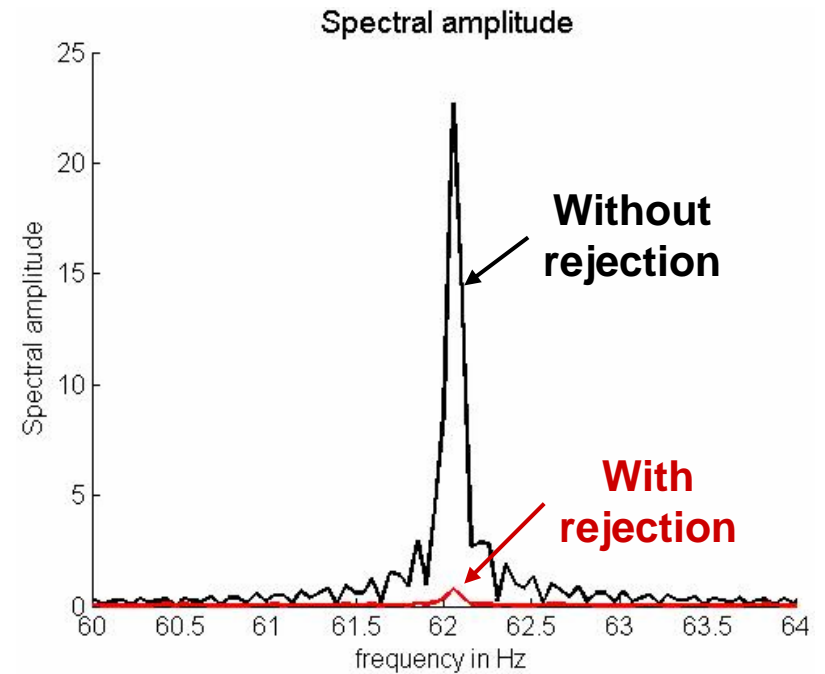


➔ The loudspeaker generates a step of frequency which simulates, for example, a change of speed of a pump near the final focus.

Experimental results



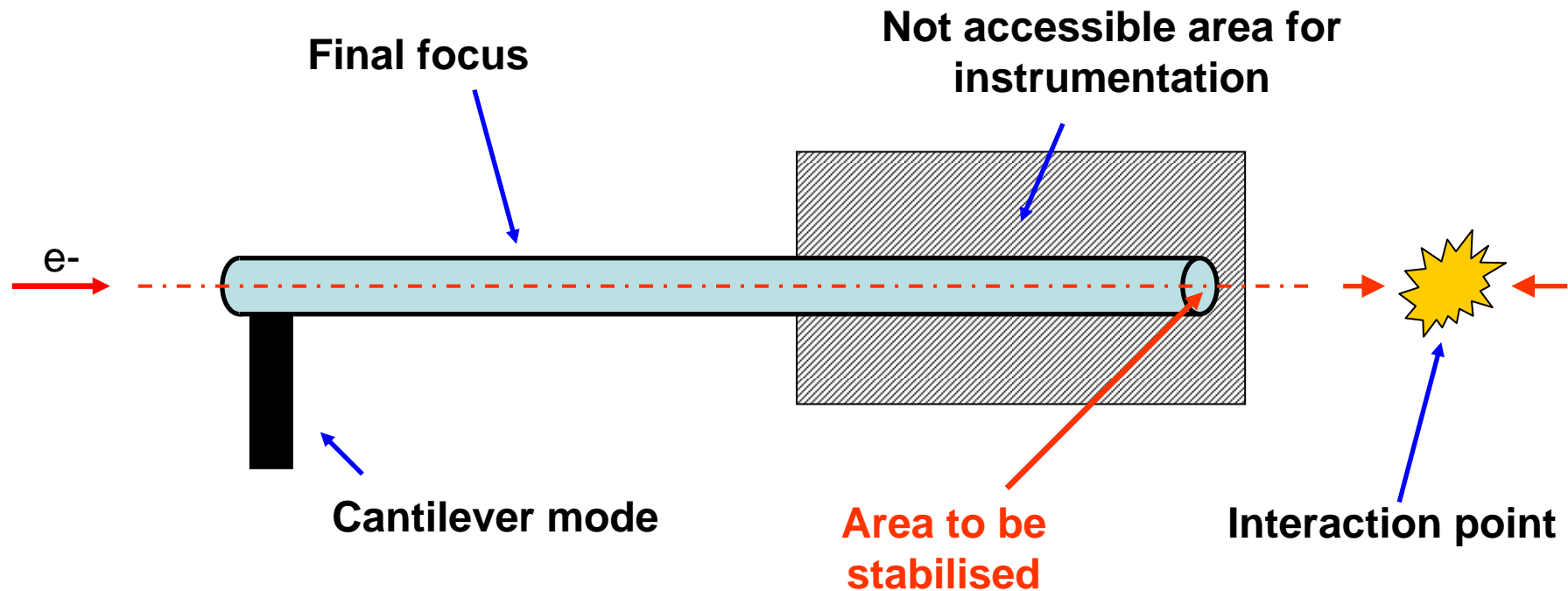
Frequency tracking in real time



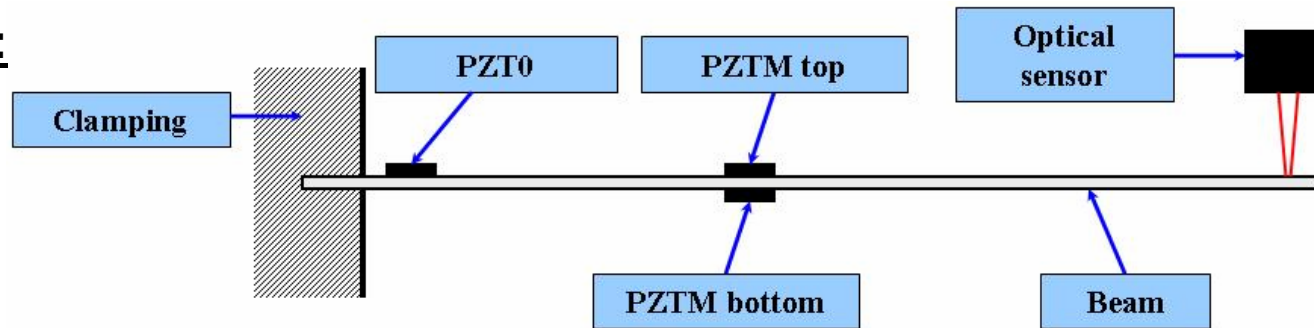
→ Efficient rejection if the variation is slow

Problem : No access to the area to be stabilised

- ➡ Where is the optimal location and what technology for the instrumentation?
- ➡ What are the effects of a local control on the global movement of the beam?
- ➡ How to be sure that the end of the beam is stabilised in real time?



The prototype :



Experimental results :

Actuator	Sensor	PZT0	PZTM	Optical
PZT0	PZTM		VG	G
PZT0	Optical		G	VG
PZTM	PZT0	VG	VB	G
PZTM	PZTM	N	VG	N
PZTM	Optical	G	VB	VG

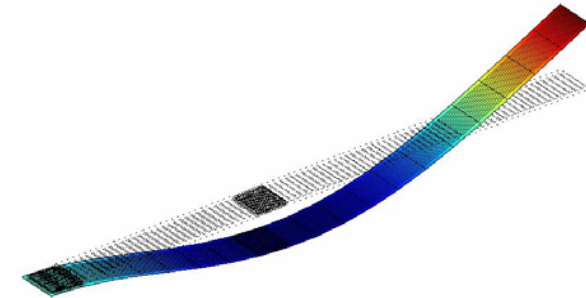
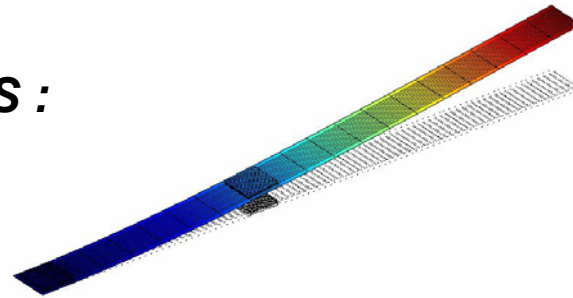
Very bad
No effect
Good
Very good

- ➔ The rejection always works at the measurement point of the feed-back.
- ➔ The behaviour of the beam changes with the configuration.

Numerical simulation of the small mock-up

Mode 1 at 12.58 Hz

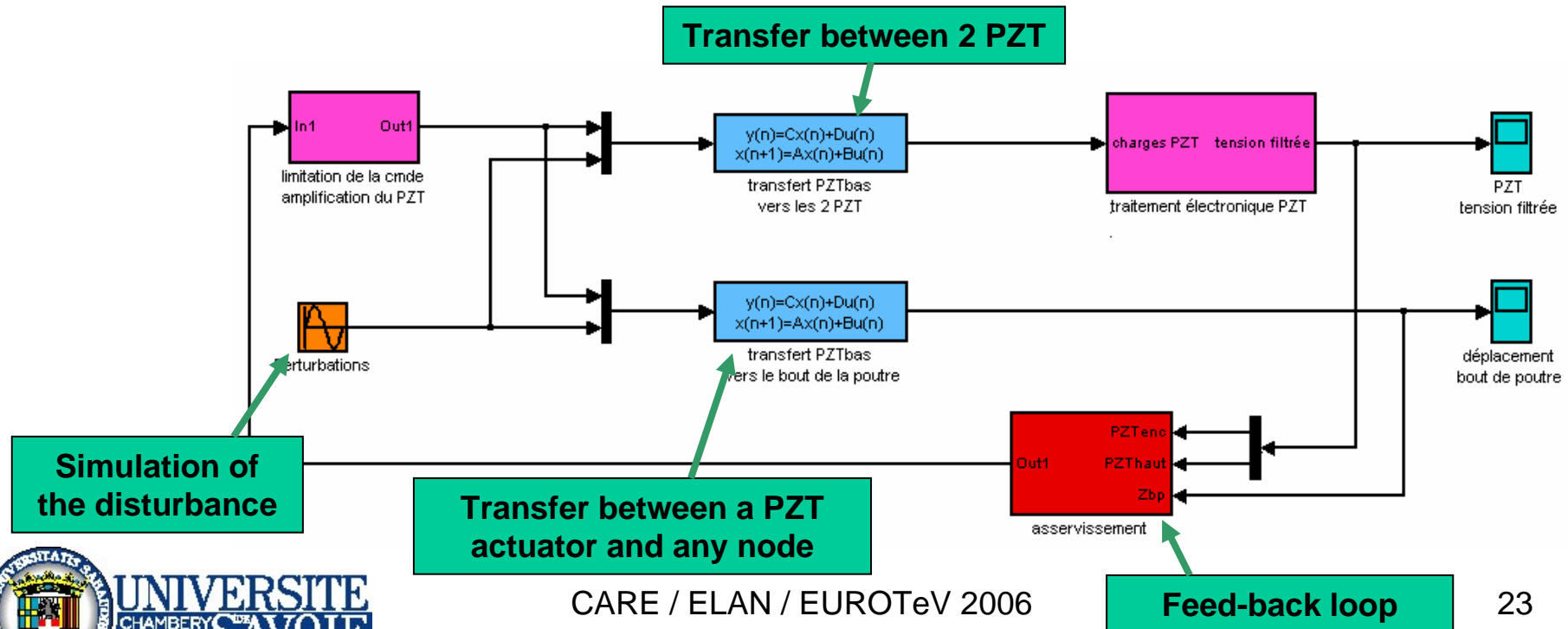
Mode 2 at 73.77 Hz



- **Finite elements with ANSYS :**

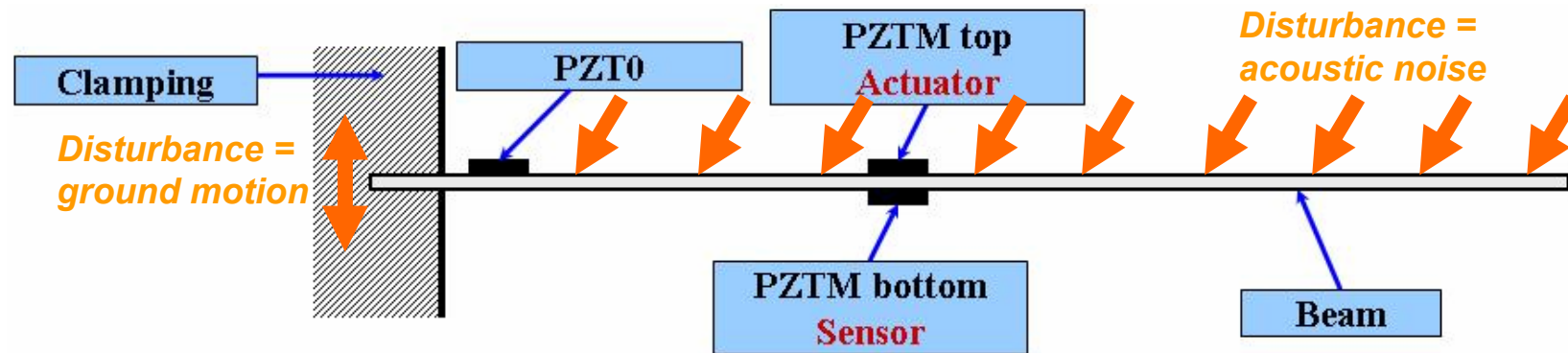
(Collaboration with LMECA)

- **“Structural Dynamic Toolbox” is used to process the characteristics of the model under Matlab / Simulink environment :**



Numerical simulation of the small mock-up

Example of the simulation :

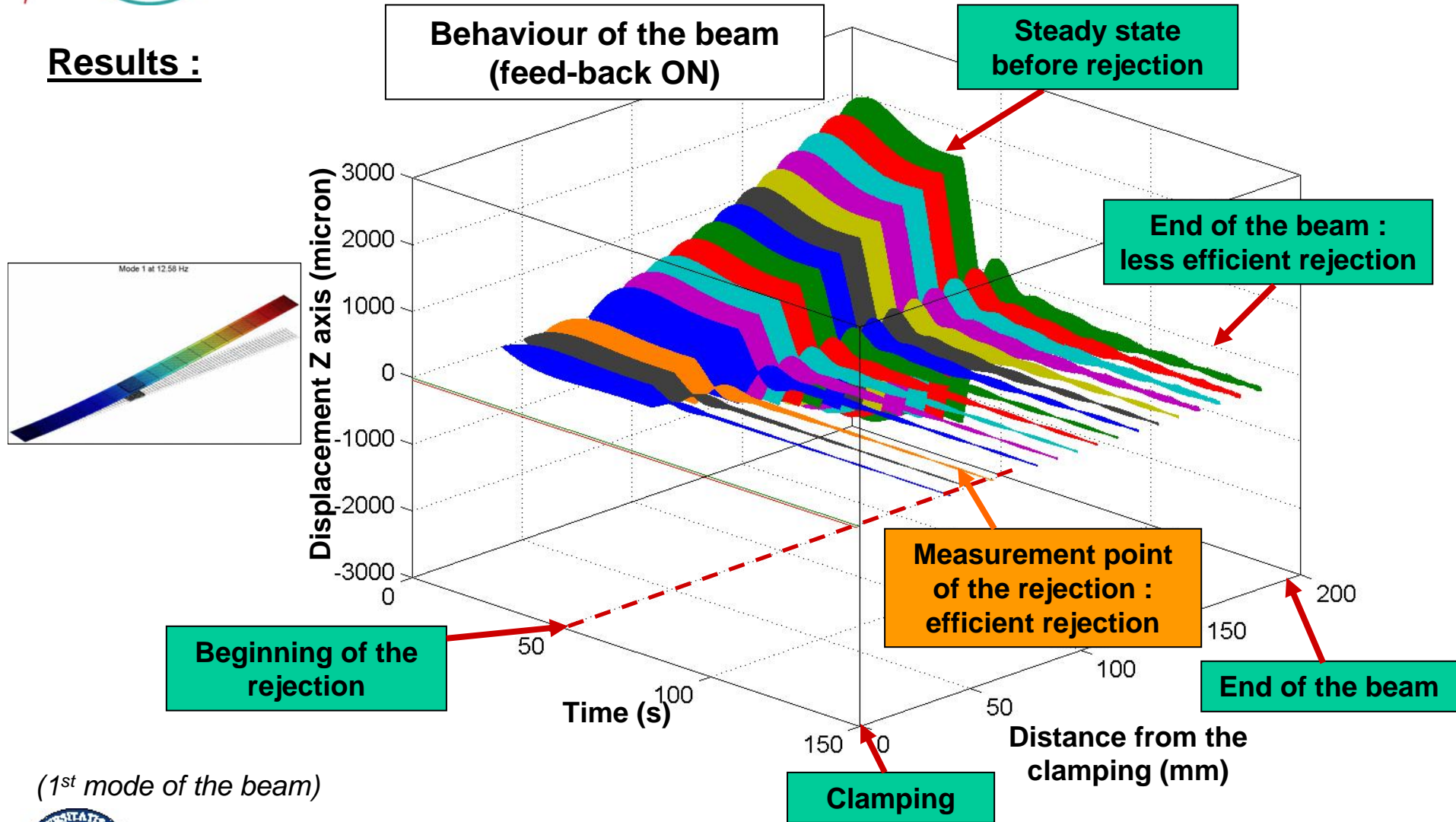


Configuration of the test :

- Rejection of one disturbance frequency (1st mode of flexion)
- “PZTM top” in actuator / “PZTM bottom” in sensor
- Monitoring simultaneously of :
 - each node of the beam
 - the voltage of each PZT patch

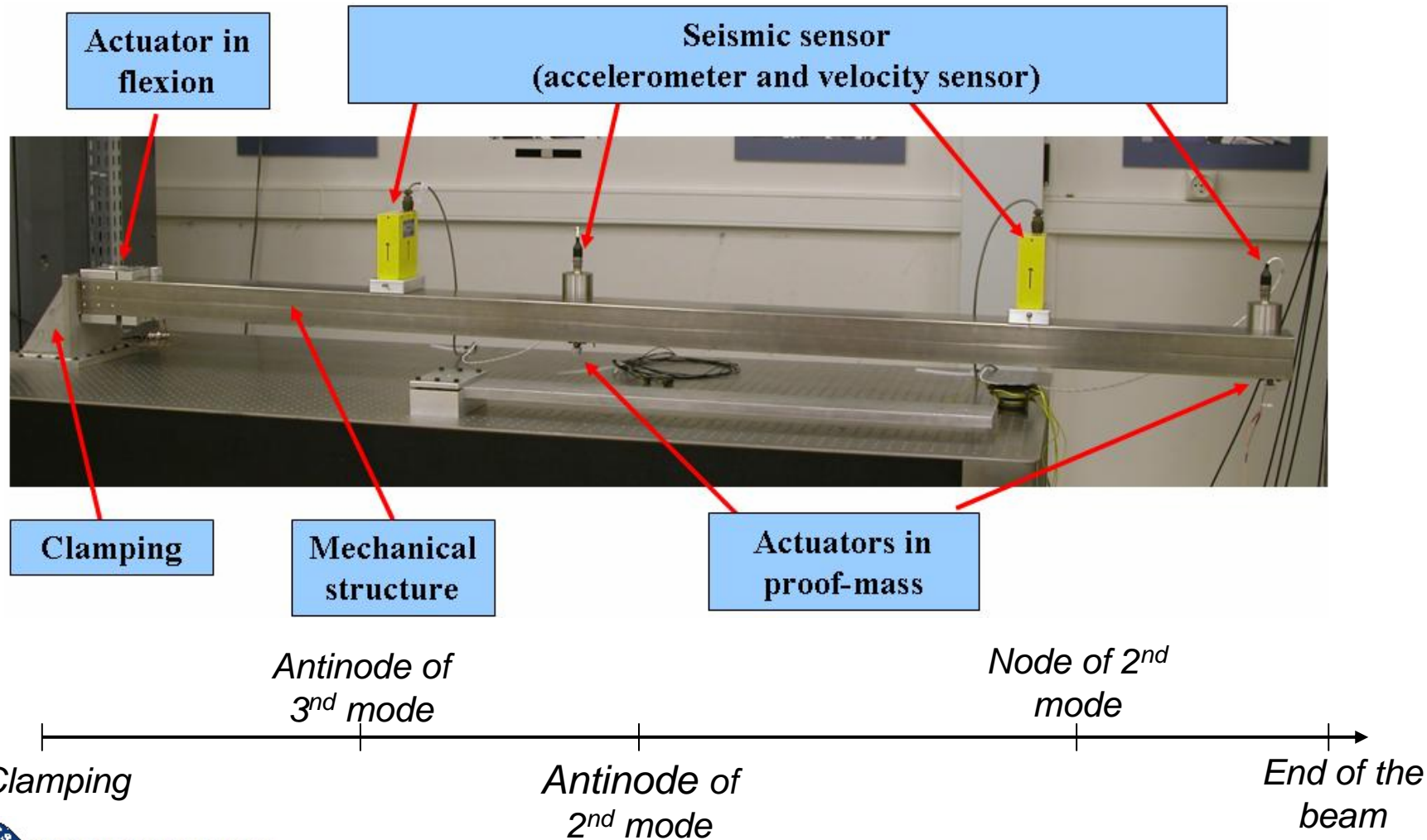
Numerical simulation of the small mock-up

Results :



(1st mode of the beam)

The prototype :



Actuator in flexion

Seismic sensor (accelerometer and velocity sensor)

Clamping

Mechanical structure

Actuators in proof-mass

Antinode of 3rd mode

Node of 2nd mode

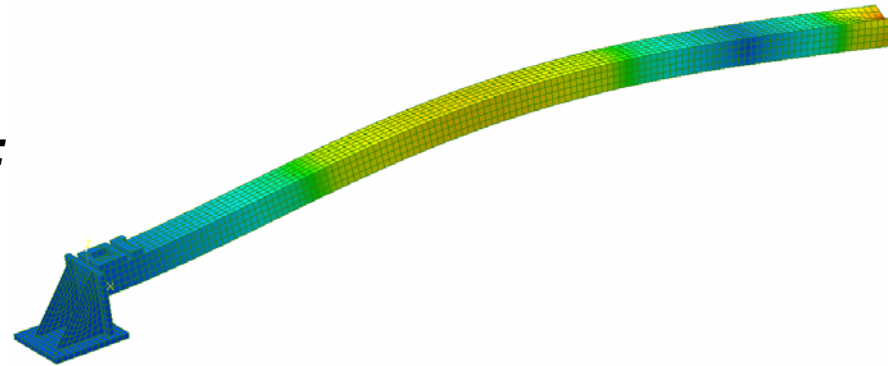
Clamping

Antinode of 2nd mode

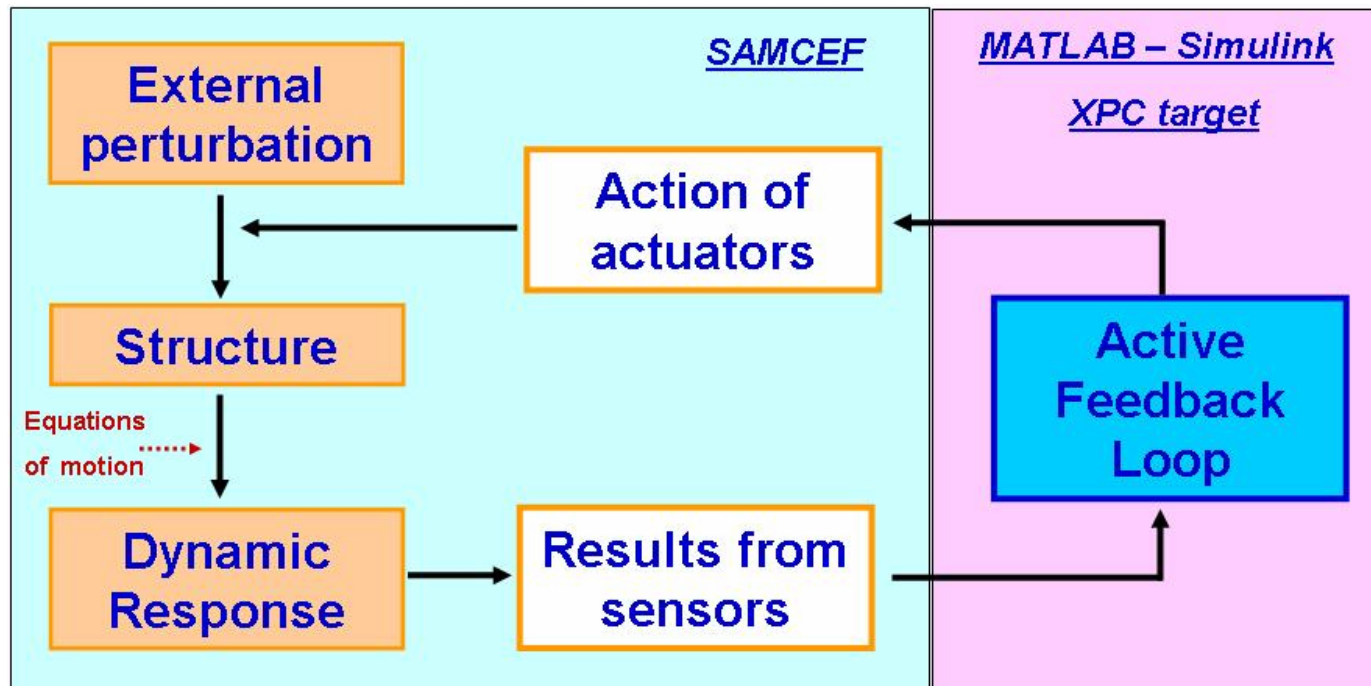
End of the beam

Simulation of the large prototype

- Finite elements with SAMCEF :



- Simulation of the entire system :



Conclusions

- Active feedback loop on a large scale prototype
 - ➔ Validation of the method for the micrometer scale
 - ➔ Validation of the frequency tracking
 - ➔ Requirement of an efficient hardware (data acquisition) to get results at nanometer scale
- Choice of the location and the technology of the instrumentation
 - ➔ First approach with experimental tests
 - ➔ Requirement of simulation for validation (with accurate updating models)
 - ➔ Multivariable problem with many sensors and actuators using different technologies