

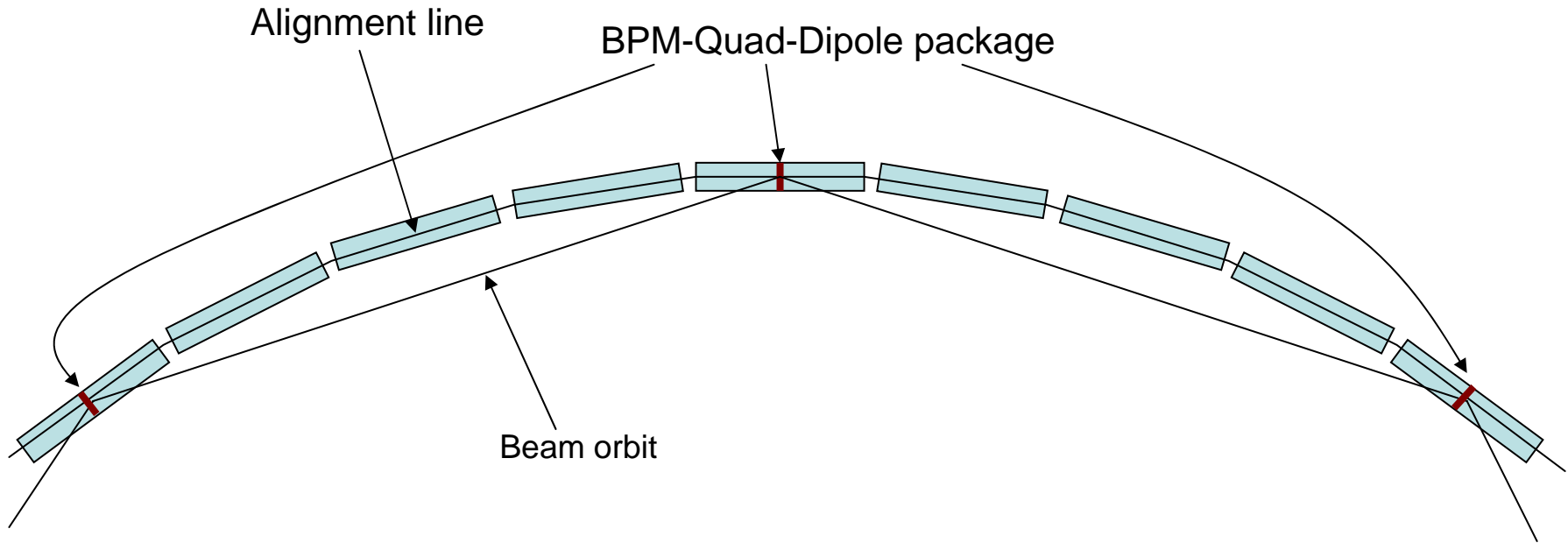
DFS(Dispersion Free Steering) in Curved(following Earth) ILC Main Linac

Kiyoshi Kubo 2006.June06

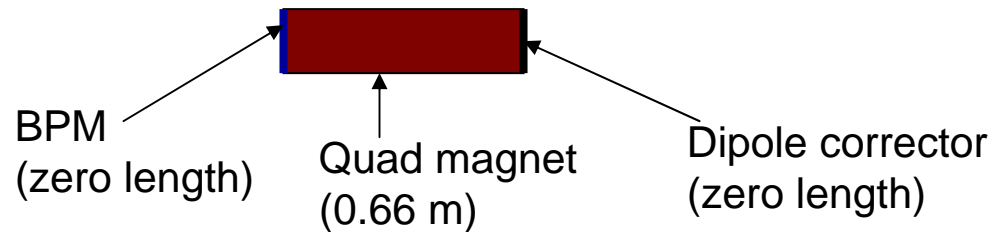
Lattice

- For Performance Study by P. Tenenbaum
 - one quad per four cryomodules
 - 75/65 degree phase advance per FODO cell

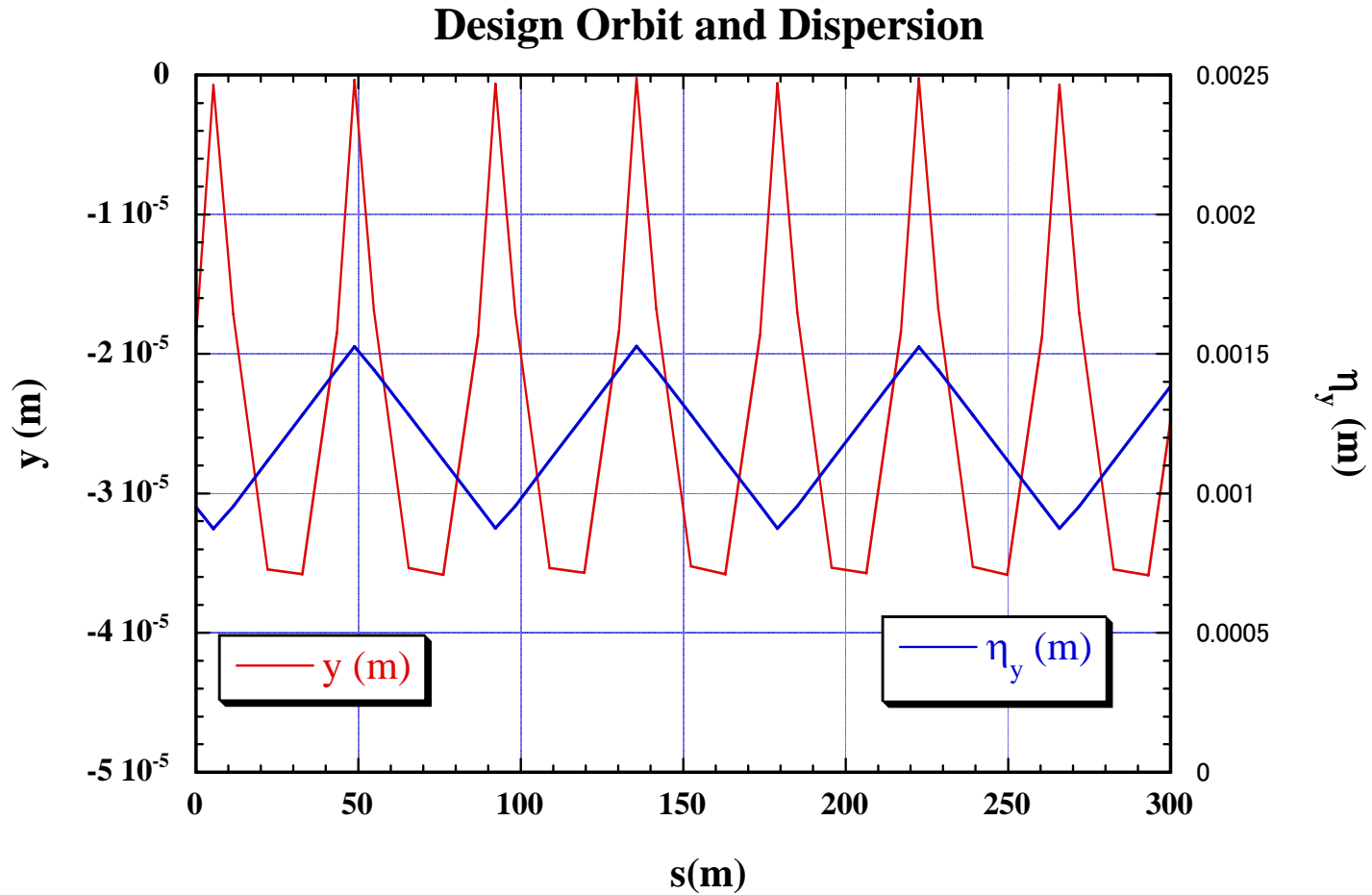
Curved Linac, 1-quad/4-cryomodules



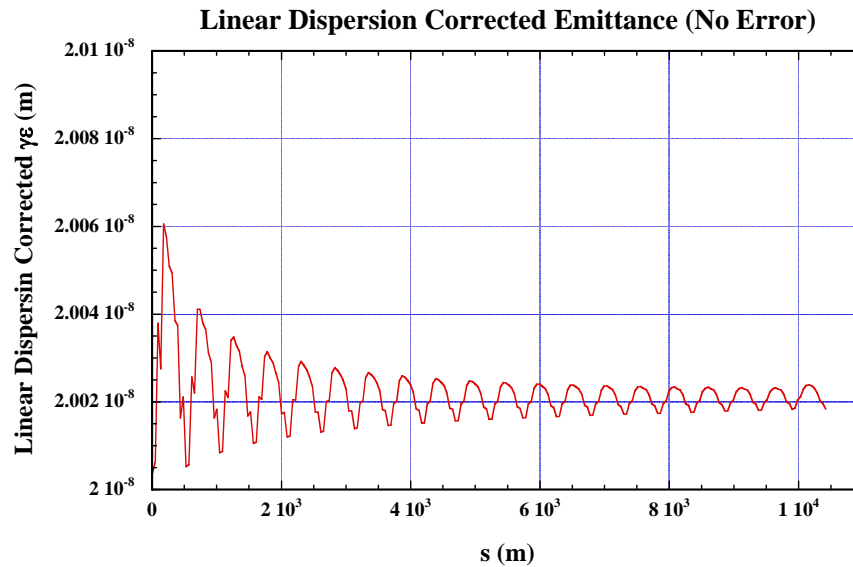
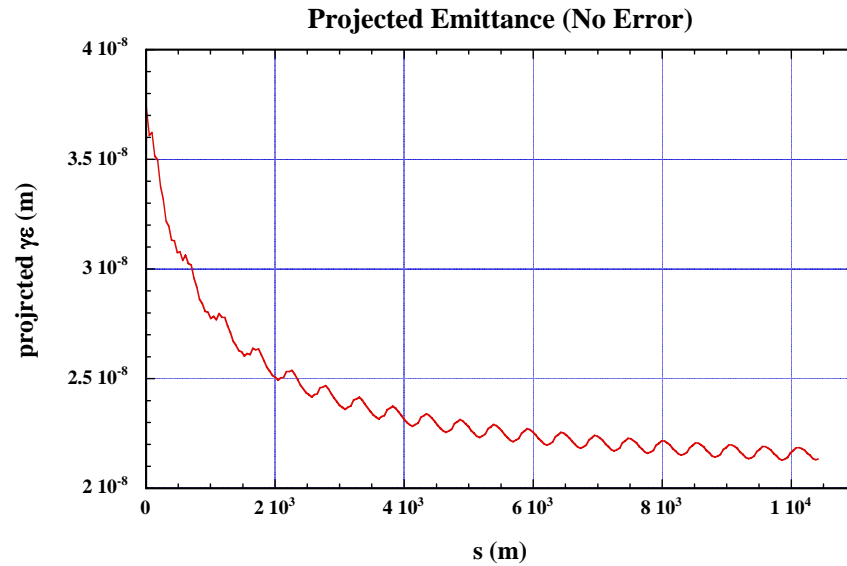
BPM-Quad-Dipole package



Design orbit and dispersion



Emittance without errors



Definition of Projected emittance and Linear Dispersion Corrected emittance

Projected emittance

$$\equiv \sqrt{(\langle y^2 \rangle - \langle y \rangle^2)(\langle y'^2 \rangle - \langle y' \rangle^2) - (\langle yy' \rangle - \langle y \rangle \langle y' \rangle)^2}$$

Linear Dispersion Corrected emittance

$$\equiv \sqrt{(\langle (y - \eta\delta)^2 \rangle - \langle y - \eta\delta \rangle^2)(\langle (y' - \eta'\delta)^2 \rangle - \langle y' - \eta'\delta \rangle^2) - (\langle (y - \eta\delta)(y' - \eta'\delta) \rangle - \langle y - \eta\delta \rangle \langle y' - \eta'\delta \rangle)^2}$$

y : Vertical offset, y' : Vertical angle

δ : Relative energy deviation

$$\eta \equiv (\langle y\delta \rangle - \langle y \rangle \langle \delta \rangle) / (\langle \delta^2 \rangle - \langle \delta \rangle^2), \quad \eta' \equiv (\langle y'\delta \rangle - \langle y' \rangle \langle \delta \rangle) / (\langle \delta^2 \rangle - \langle \delta \rangle^2)$$

$\langle \rangle$: Average over all macro - particles

Simulated Algorithm of DFS, mode 0

One-to-one orbit correction (BPM reading zeroed)

Divide linac into sections (can be overlapped) and in each section:

(1) Measure orbit with nominal beam energy. ($y_{0,i}$ at i-th BPM)

(2) Reduce initial beam energy and accelerating gradient in entire linac by a common factor δ (e.g. 10% or $\delta = -0.1$).

(3) For the second section or downstream, orbit adjusted at the two BPMs just before the section to make the position at the BPM

$$y_{\delta} = y_0 + \delta\eta$$

(y_0 is the position with nominal energy, η the dispersion at BPM.)

(4) Measure orbit. ($y_{\delta,i}$ at i-th BPM)

(5) Set dipole correctors in the section to minimize

$$w\Sigma(y_{\delta,i} - y_{0,i} - \delta\eta_i)^2 + \Sigma(y_{0,i} - y_{des,i})^2$$

(η_i is the dispersion, $y_{des,i}$ the designed orbit at i-th BPM. w is the weight factor, chosen as $w=5000$.)

(6) Iterate from (1) to (5).

(7) Go to next section.

Simulated Algorithm of DFS, mode 1

One-to-one orbit correction (BPM reading zeroed)

Divide linac into sections (can be overlapped) and in each section:

(1) Measure orbit with nominal beam energy. ($y_{0,i}$ at i-th BPM)

(2) Reduce initial beam energy and accelerating gradient from the linac entrance to the entrance of the section by a common factor δ (e.g. 10% or $\delta = -0.1$).

(3) For the second section or downstream, orbit adjusted at the two BPMs just before the section to make the position at the BPM

$$y_{\delta} = y_0 + \delta\eta$$

(y_0 is the position with nominal energy, η the dispersion at BPM.)

(4) Measure orbit. ($y_{\delta,i}$ at i-th BPM)

(5) Set dipole correctors in the section to minimize

$$w\Sigma(y_{\delta,i} - y_{0,i} - \Delta y_{\text{cal},i})^2 + \Sigma(y_{0,i} - y_{\text{cal},i})^2$$

($\Delta y_{\text{cal},i}$ is the calculated orbit difference, $y_{\text{cal},i}$ the calculated orbit, without errors, at i-th BPM. w is the weight factor, $w=5000$.)

(6) Iterate from (1) to (5).

(7) Go to next section.

Simulated Algorithm of DFS, mode 2

One-to-one orbit correction (BPM reading zeroed)

Divide linac into sections (can be overlapped) and in each section:

(1) Measure orbit with nominal beam energy. ($y_{0,i}$ at i-th BPM)

(2) Reduce initial beam energy and accelerating gradient from the linac entrance to the entrance of the section by a common factor δ (e.g. 10% or $\delta = -0.1$).

(3) (No upstream orbit adjustment)

(4) Measure orbit. ($y_{\delta,i}$ at i-th BPM)

(5) Set dipole correctors in the section to minimize

$$w \sum (y_{\delta,i} - y_{0,i} - \Delta y_{\text{cal},i})^2 + \sum (y_{0,i} - y_{\text{cal},i})^2$$

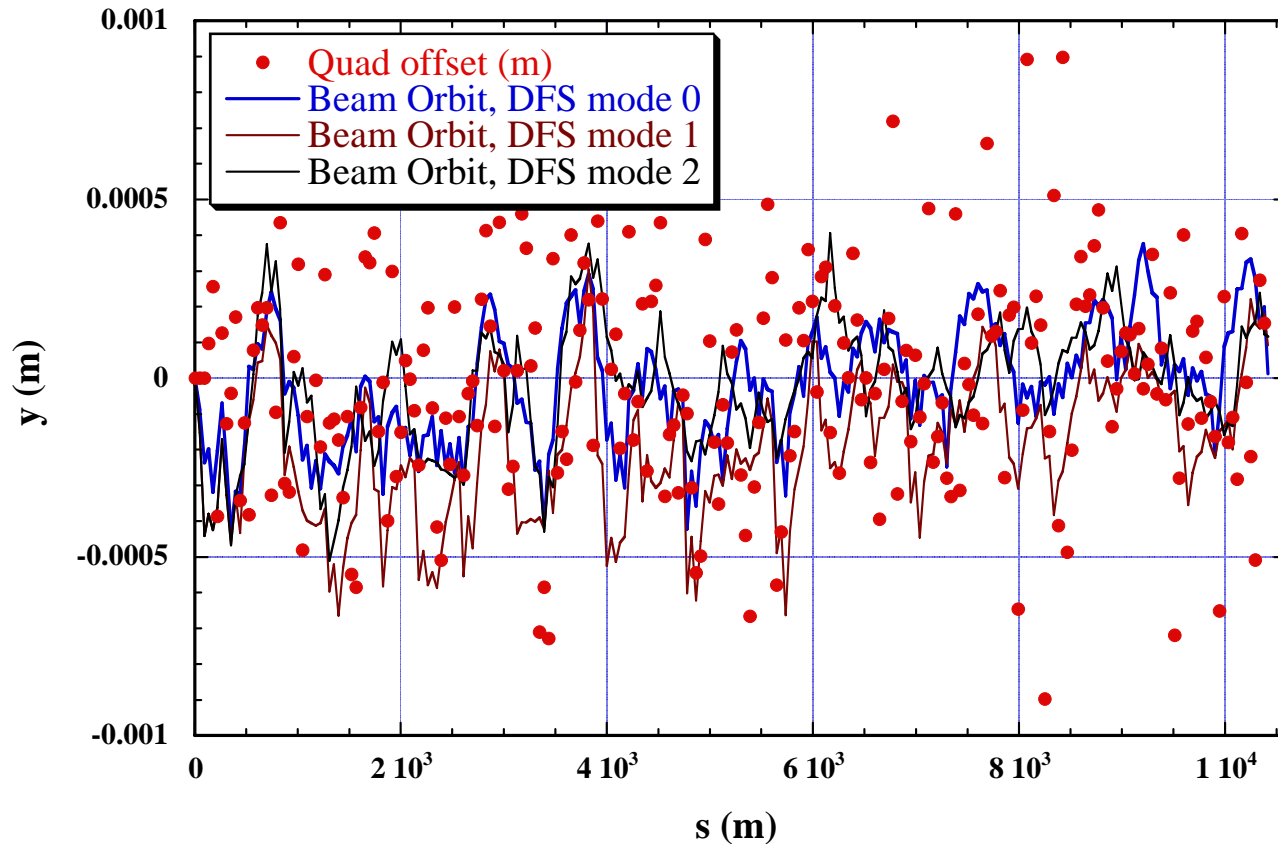
($\Delta y_{\text{cal},i}$ is the calculated orbit difference, $y_{\text{cal},i}$ the calculated orbit, without errors, at i-th BPM. w is the weight factor, $w=5000$.)

(6) Iterate from (1) to (5).

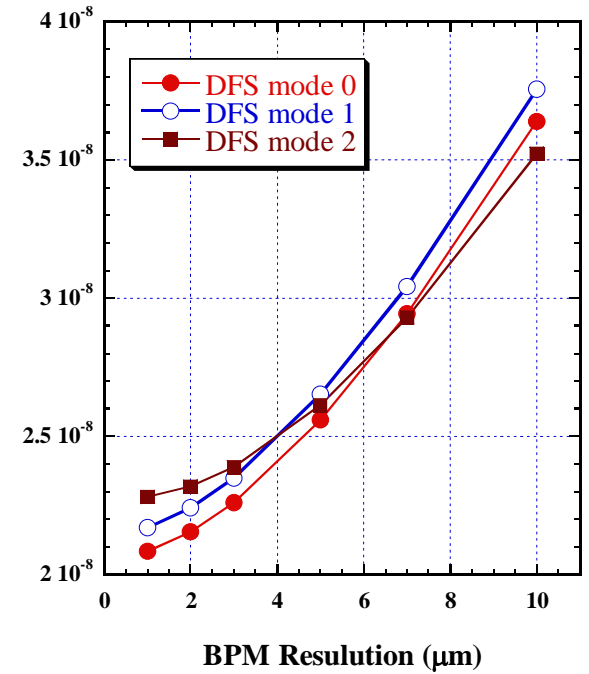
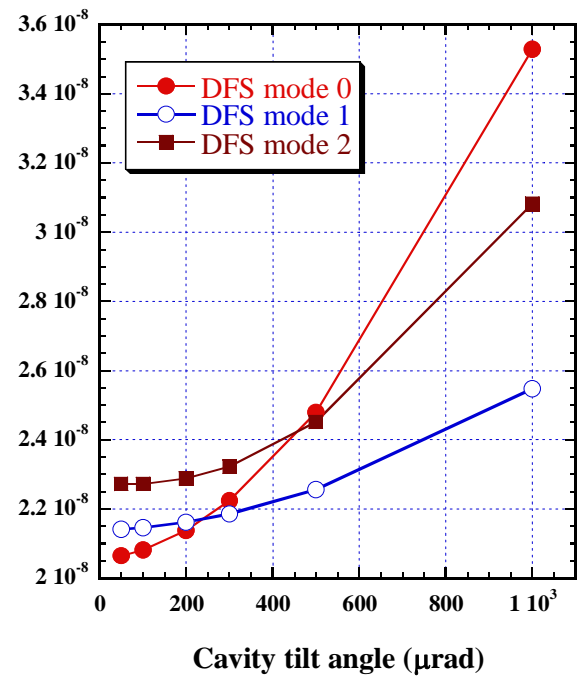
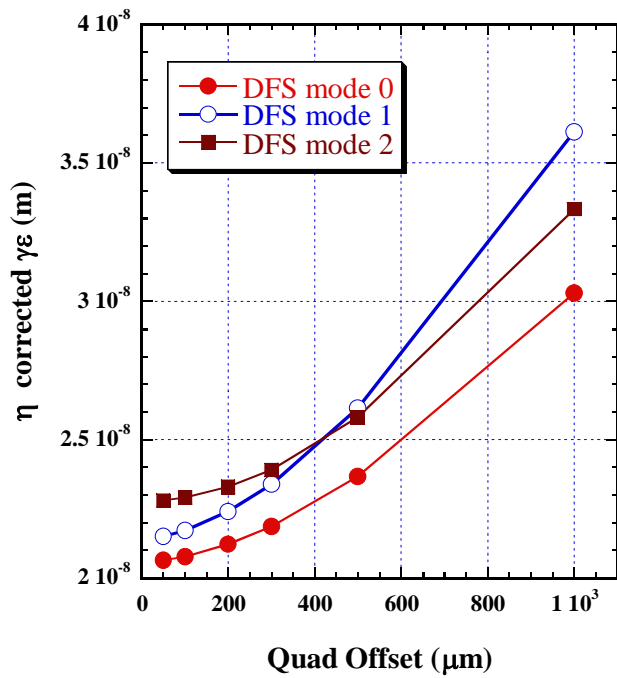
(7) Go to next section.

Example of beam orbit after DFS

Quad offset 0.3 mm, Cavity offset 0.5 mm, Quad-BPM offset 0.2 mm, Cavity tilt 0.5 mrad, BPM resolution 3 micron. (example of one particular seed.)

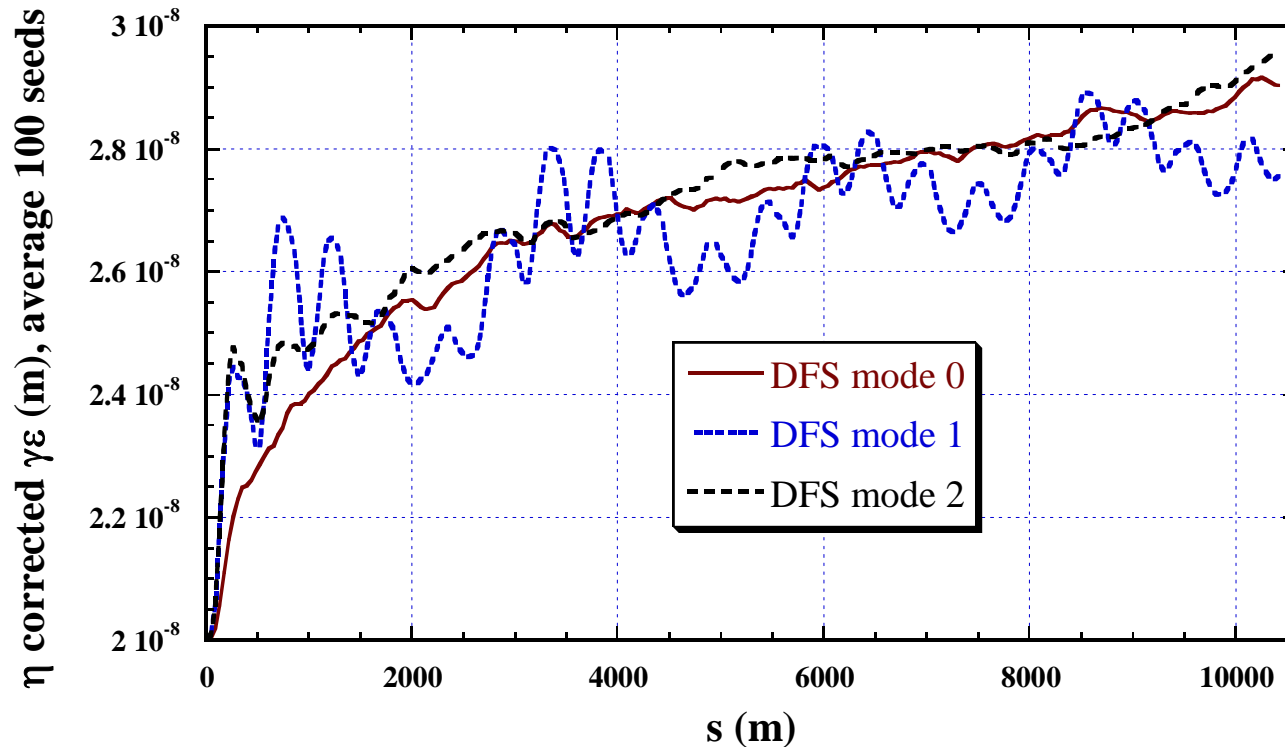


Sensitivity of final (linear dispersion corrected) emittance to various errors (Average of 25 random seeds. No other errors in each figure.)



Emittance vs. s

Quad offset 0.3 mm, Cavity offset 0.5 mm, Quad-BPM offset 0.2 mm, Cavity tilt 0.5 mrad, BPM resolution 3 micron. (Average of 100 random seeds.)



Mode 1 shows clear structure due to division into sections.

Summary and comments

- DFS (Dispersion Free Steering) is effective in slowly curved (following earth) linac.
- Three different algorithms of DFS were tested.
 - Result of mode 0 is most sensitive to cavity tilt error, as expected.
 - Averaged emittance of the three looked similar for the combined errors: Quad offset 0.3 mm, Cavity offset 0.5 mm, Quad-BPM offset 0.2 mm, Cavity tilt 0.5 mrad, BPM resolution 3 micron.
 - About 40% to 50% vertical emittance dilution. (w.r.t. $2E-8$ m)
- The results seem to depend on how the linac is divided into sections. (not shown here)